

DESIGN AND WIND TUNNEL TESTS OF AN AIRFOIL FOR THE HORIZONTAL TAILPLANE OF A STANDARD CLASS GLIDER

In the above paper, by L.M.M. Boermans and F. Bennis, published in Volume 16, No. 2, the table of ordinates was incomplete. The complete table for this paper is as follows.

TABLE 1. Coordinates of DU86 - 137/25

100.000	0.000	44.559	6.790	.047	-.301	50.084	-6.582
99.871	.014	41.606	6.846	.294	-.696	53.040	-6.426
99.492	.056	38.678	6.864	.764	-1.091	55.980	-6.228
98.880	.124	35.786	6.844	1.435	-1.502	58.894	-5.984
98.043	.219	32.941	6.788	2.290	-1.925	61.773	-5.691
96.988	.341	30.155	6.697	3.318	-2.352	64.612	-5.347
95.722	.488	27.439	6.571	4.516	-2.770	67.410	-4.950
94.254	.661	24.803	6.411	5.882	-3.181	70.167	-4.506
92.592	.860	22.256	6.217	7.410	-3.583	72.880	-4.019
90.745	1.084	19.809	5.992	9.093	-3.972	75.554	-3.500
88.721	1.332	17.471	5.735	10.924	-4.348	78.191	-2.970
86.531	1.605	15.249	5.448	12.894	-4.708	80.775	-2.457
84.183	1.902	13.152	5.132	14.994	-5.048	83.286	-1.981
81.686	2.223	11.186	4.791	17.216	-5.364	85.696	-1.553
79.037	2.571	9.359	4.425	19.552	-5.655	87.982	-1.179
76.229	2.984	7.677	4.039	21.944	-5.918	90.121	-.862
73.352	3.541	6.146	3.636	24.533	-6.151	92.092	-.604
70.546	4.177	4.774	3.218	27.157	-6.351	93.875	-.398
67.796	4.723	3.565	2.788	29.857	-6.517	95.455	-.242
65.012	5.168	2.524	2.349	32.625	-6.646	96.821	-.128
62.179	5.555	1.656	1.904	35.450	-6.736	97.963	-.058
59.305	5.881	.964	1.457	38.323	-6.788	98.855	-.030
56.394	6.154	.453	1.015	41.232	-6.799	99.488	-.017
53.454	6.379	.138	.578	44.169	-6.768	99.870	-.004
50.494	6.558	.007	.131	47.124	-6.695	100.000	0.000
47.526	6.695						